

CONVAIR ASTRONAUTICS

CONVAIR DIVISION OF GENERAL DYNAMICS CORPORATION

MECHANICAL PROPERTIES OF
ADLOCK 851 AT ROOM TEMPERATURE,
1000°, -320° AND -423°F

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PREPARED BY **L. F. Nelson**
Engineering Metallurgist
CHECKED BY **F. P. Brodell**
F. P. Brodell
Senior Engineering Metallurgist

APPROVED BY **W. M. Gross**
Test Lab Group Engineer
APPROVED BY **W. S. Campbell**
Chief of Test Labs

REVISIONS

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MECHANICAL PROPERTIES OF ADLOCK 851
AT ROOM TEMPERATURE, 1000°, -320° AND -423°F

OBJECT:

To determine the tensile and flexural properties of Adlock 851 fiberglass laminate at room temperature, 1000°, -320° and -423°F.

CONCLUSIONS:

Flexural Test. The maximum fiber stress for this particular cure of Adlock 851 increased from 65.4 ksi at 78° to 84.3 ksi at -320°F. Fiber stress values then decreased from 84.3 ksi at -320°F to 81.0 ksi at -423°F. Deflection values to failure also follow the same trend, i.e., increasing from 0.121 inches at 1000°F to 0.149 inches at -320°F and then decreasing to 0.117 inches at -423°F. Modulus of elasticity values increase consistently with decreasing temperature from 2.92×10^6 at 78° to 3.49×10^6 at -423°F.

Tensile Properties. The tensile properties of Adlock 851 follow along the same general lines as the flexural test results. The ultimate strength increased from 31.5 ksi at 1000°F to 73.6 ksi at -320°F. From -320°F to -423°F, the ultimate strength decreased from 73.6 ksi to 63.6 ksi.

SPECIMENS:

The tensile specimens were the standard 9 inch coupons machined according to Structural Standard 001 (Figure 1).

Flexural specimens were machined according to Figure 2.

PROCEDURE:

Flexure specimens tested at 1000°F were heated to this temperature on one side only. It took 90 seconds to reach 1000°F from ambient.

A static load of 6 ksi was placed on the flexure specimens prior to heating. Photographs of the test set up are shown in Figures 5 and 6. Heat was applied to the flexure specimens with a series of quartz heat lamps, whose output was controlled by a 115V variable autotransformer. Temperature readings were followed by attaching a thermocouple to the heated side of the specimen and recording millivolt output on a Sanborn 150 Recorder. Deflection values were determined by using a linear transducer in conjunction with a Sanborn 150 Recorder. Maximum deflection values, at failure, were recorded at the highest temperature reached.

Flexural tests at the other temperatures used the same test set up but the environment was changed to room temperature, liquid nitrogen and liquid hydrogen for the other reported temperature.

From the recorded values, maximum fiber stress and modulus of elasticity in bending was calculated. The following formulas were used to calculate maximum fiber stress and the modulus of elasticity in bending respectively:

$$(1) \quad S = \frac{3PL}{2bd^2}$$

Where S = maximum fiber stress psi
P = maximum load in pounds
L = distance between supports, inches
b = width of beam tested, inches
d = depth of beam tested, inches

$$(2) \quad E_B = \frac{L^3}{4bd^3} (P/Y)$$

Where E_B = modulus of elasticity in bending

L = distance between supports, inches
b = width of beam tested, inches
d = depth of beam as tested, inches
P/Y = slope of straight line portion of load deflection curve in pounds/inch

Tensile specimens were tested at 1000°F, ambient, -320°F and -423°F. Stress strain curves were run on each specimen.

RESULTS AND DISCUSSION:

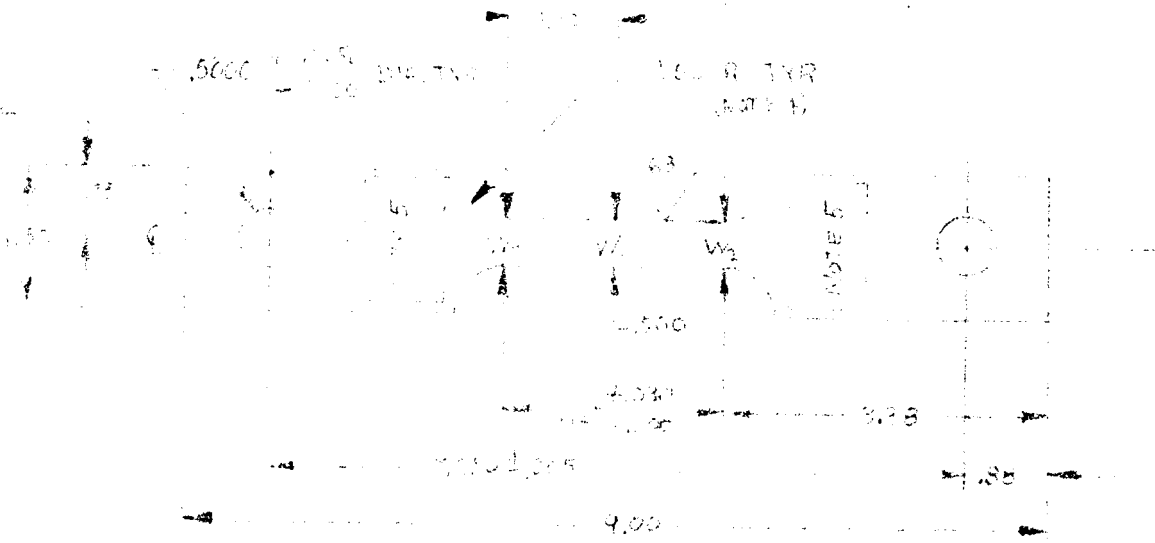
The results of the flexural and tensile data at 1000°F, room temperature, -320°F and -423°F are given in Figures 3 and 4. The values reported for the tensile test are open to question for the following reasons:

1. Many of the specimens failed in the end fitting at 1000°F.
2. All of the ambient tensile specimens failed in the radius and not in the center of the test area.
3. Failure at cryogenic temperatures was either in the end fitting, in the radius, or was initiated by the knife edges which were used to obtain stress strain data.

The tensile data is recorded in Figure 4.

PHOTO INDEX

<u>Figure No.</u>	<u>Photo No.</u>	<u>Title</u>	<u>Page</u>
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**NOTES**

1. Holes on centerline of test section within $\pm .005$.
2. Gradual taper from W_2 to W_1 of $.004 \pm .001$ in. W_2 to be greater than W_1 .
3. Test section to have sharp corners free from burrs.
4. No undercut at intersection of radius and test section.
5. Identify here by electro etch with Heat No., Coil No., Specification No., and Specimen No.

STRUCTURES STANDARD 001		CONVAIR SYSTEMS	
DOUBLE-FLANGE COUPON		MATERIALS	
		LABORATORY	
		AAL-274	
		SHEET 1 OF 1	

FLEXURAL SPECIMEN

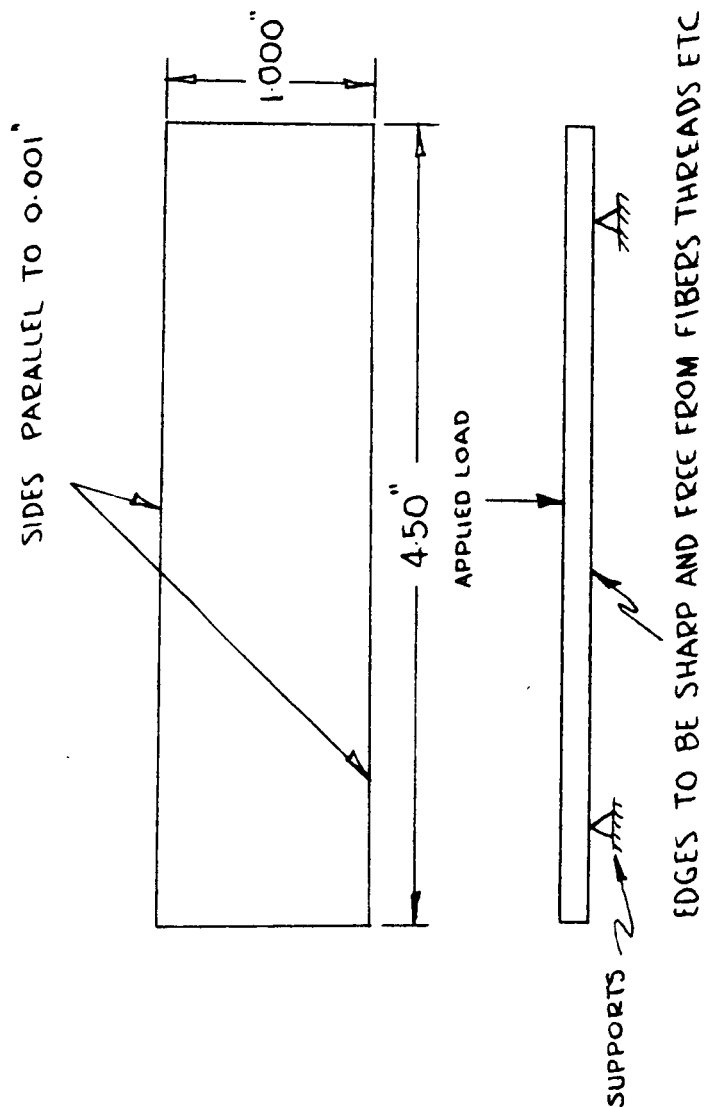


FIG 2

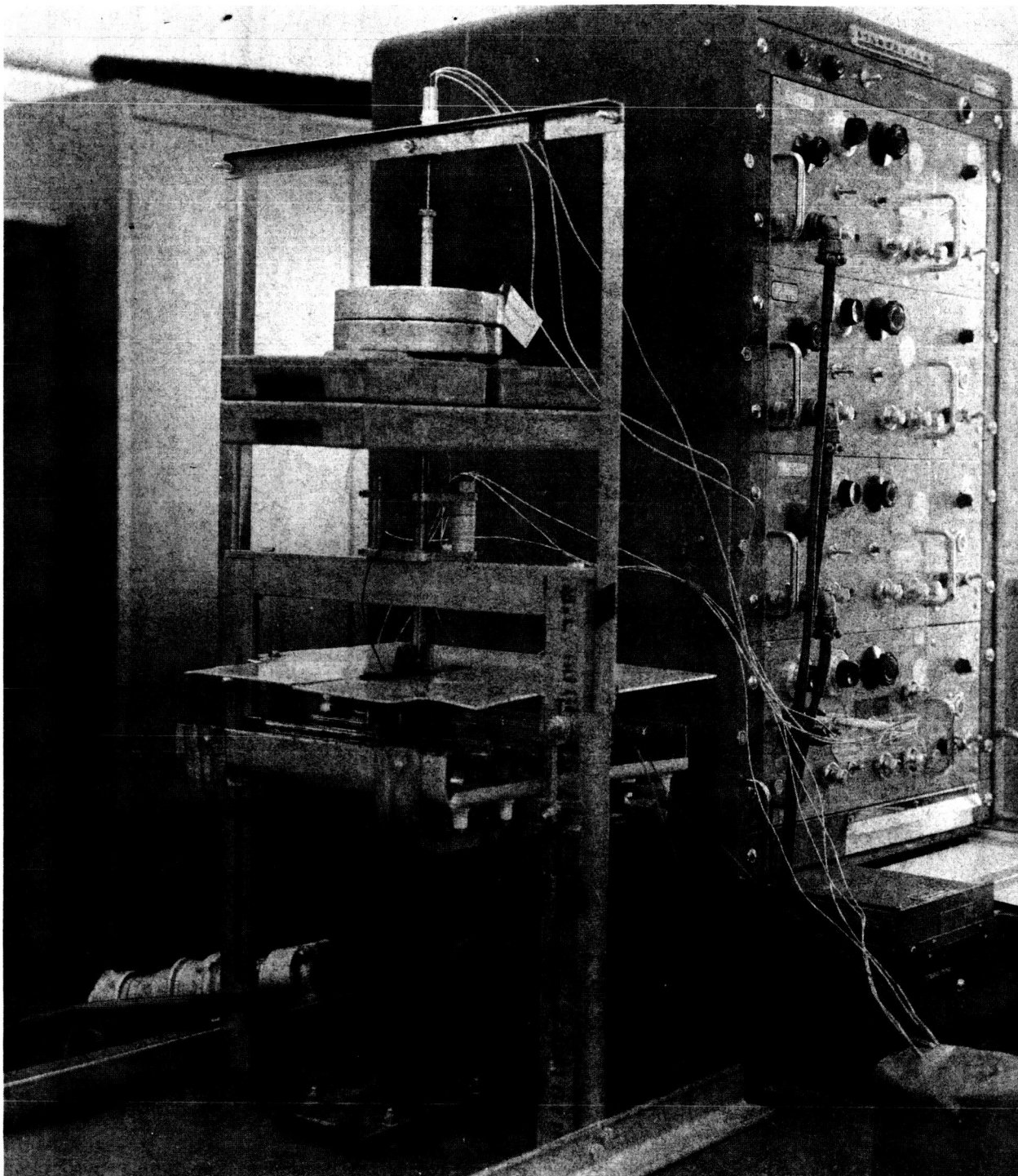
FORM A-423

FIG 3 - ADLOCK 851 - FLEXURAL PROPERTIES OF

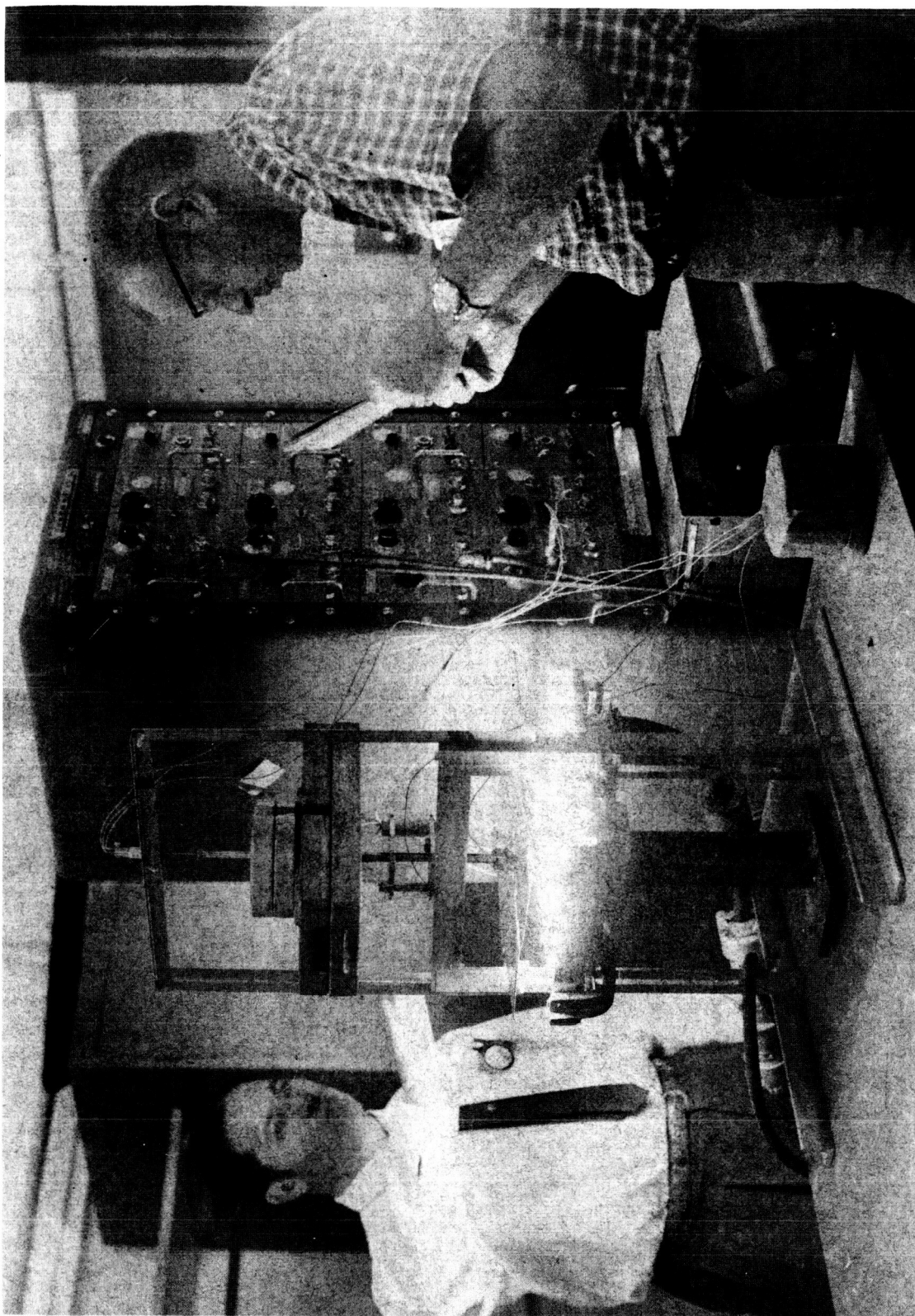
FORM A-423

FIG 3 - ADLOCK 851- FLEXURAL PROPERTIES OF

[illegible]



FLEXURE TEST SET UP FOR USE AT 1000°F UNDER CONSTANT LOAD
FIGURE NO. 5



ACTUAL TEST RUN OF FLEXURE TEST AT 1000°F
FIGURE NO. 6